

Materializing the Vision of “Flying Carpets”

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www.pes.ee.ethz.ch

June 14, 2023



Ultra-Lightweight/Efficient Power Electronics Enabling Future Urban Air Mobility

Johann W. Kolar, **David Menzi**, **Luc Imperiali**, **Elias Bürgisser**, **Jonas E. Huber**



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Materializing the Vision of “Flying Carpets” – Ultra-Lightweight/Efficient Power Electronics Enabling Future Urban Air Mobility

Urban Air Mobility (UAM) based on electric vertical take-off and landing (eVTOL) aircraft – a 21st century materialization of the legendary “Magic/Flying Carpet” – are based on multi-rotor or tilt-rotor/duct/wing designs, can carry four to six occupants and operate from vertiports without runways. Compared to terrestrial alternatives this allows for a two- to six-fold faster means of point-to-point mobility. Aircraft electrification enables considerably higher overall efficiency as a larger number of small high-efficiency electric motors, i.e., distributed electric propulsion, can be used instead of conventional low-efficiency combustion-based propulsion architectures with few large units, resulting in reduced drag and greater flexibility to leverage the benefits of aero-propulsive coupling. Accordingly, urban and regional eVTOL aerial travel services are expected to massively expand over the next decades.

The talk first introduces key eVTOL aircraft designs currently in the R&D, prototyping, or production planning phases, discusses trade-offs of key performance indicators like range and payload using first-order principles, and highlights critical enabling technologies like high gravimetric energy density and/or high-power-density batteries and fuel cells, low-specific-weight electric motors, and advanced power electronics. Hybrid battery/fuel cell power supplies of eVTOL aircraft enable high peak power capability as well as long-range operation. However, the typically wide and overlapping voltage ranges of the batteries and the fuel cells require interconnecting bidirectional DC-DC converters with buck-boost capability.

Accordingly, the second part of the presentation comparatively evaluates performance limits of fully soft-switched, flying-capacitor-multilevel, and partial-power-processing buck-boost candidate converter topologies by means of comprehensive Pareto optimizations considering mission profile efficiency and gravimetric power density, and finally presents a 15kW 450V...730V / 480V...800V three-level flying capacitor converter module of a 150kW system featuring 98.5% efficiency and an unprecedented gravimetric power density of 62kW/kg.

Finally, a summary of first assessments of the primary energy and Greenhouse Gas Emissions impacts of eVTOLs vs. ground-based light-duty vehicles for passenger mobility is presented, which surprisingly indicates partly higher energy efficiencies than equivalent terrestrial alternatives at faster and more predictable travel times, and indicates a possible niche role of eVTOLs in future sustainable urban transportation.

Outline



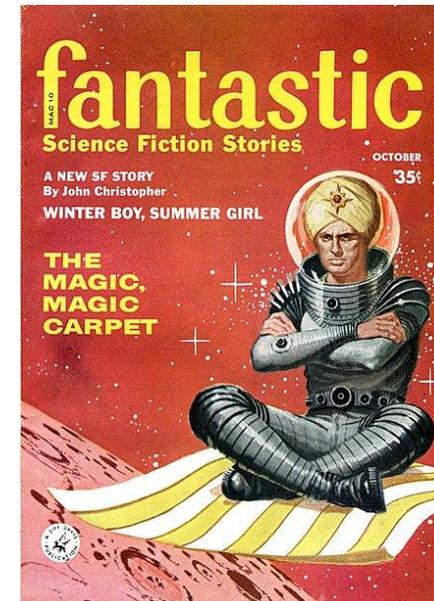
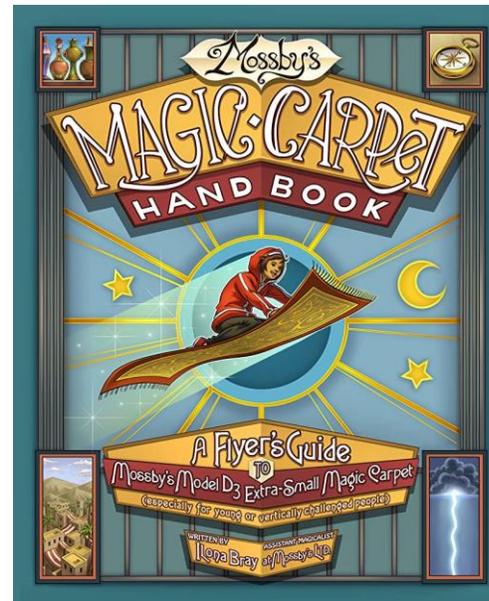
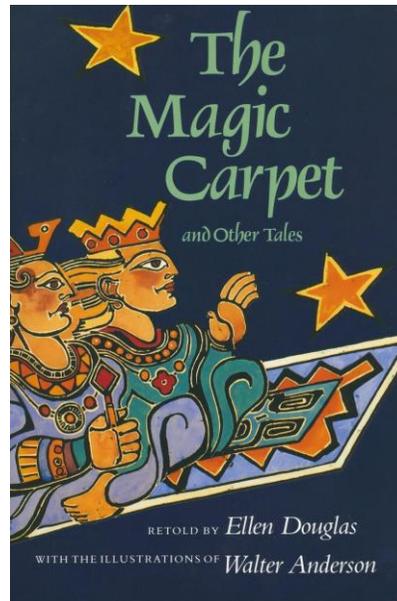
- ▶ *Once Upon a Time ...*
 - ▶ *eVTOL Aircraft Concepts*
 - ▶ *Energy / Power Sources*
 - ▶ *El. Motor Technologies*
 - ▶ *On-Board Power Electronics*
 - ▶ *Ultra-Fast Battery Charging*
- ▶ *Sustainability*

Acknowledgment

Martin Ulmer

Once Upon a Time ...

- “Magic Carpets” — Featured in the “1001 Nights” and Modern Literature
- Quietly and Swiftly / Instantaneously Carrying their Users to Desired Destinations



- Handbook on How to Operate a Magic Carpet for “Young or Vertically Challenged People”

Today's Motivation

- 2015 — Typ. **San Francisco** Resident Spent 230h/ Year Commuting btw. Work & Home
- **500'000** Hours of Productivity Lost / **Single Day**



Source: <http://billoodevelopment.com>



- Use **3D-Airspace** to Alleviate Transportation Congestion on the Ground — **“Flying Cars”**

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Source: <https://www.youtube.com/watch?v=44bSw-wPW4c>



- Use **3D-Airspace** to Alleviate Transportation Congestion on the Ground — **"Flying Cars"**

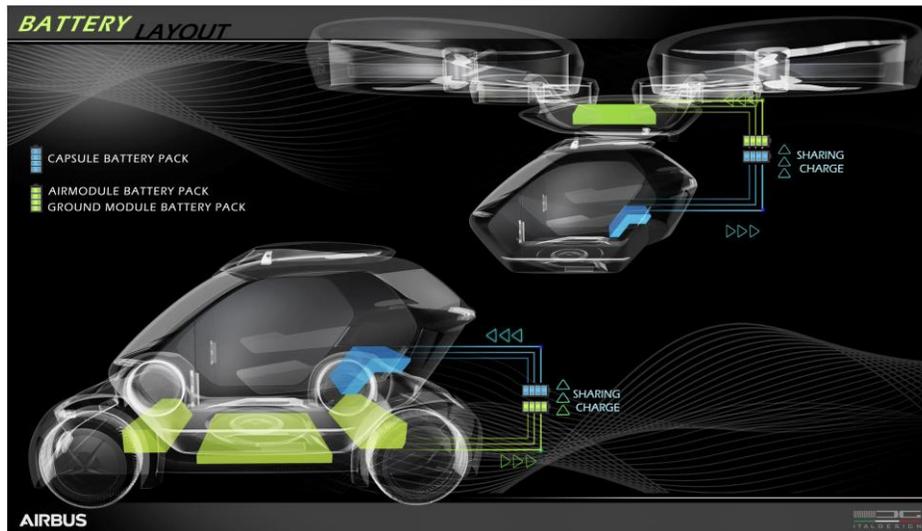
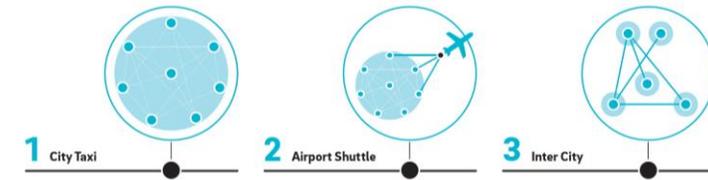
Urban Air Mobility

— *Operation Characteristic* —
eVTOL Aircraft Types

Urban Air Mobility (UAM)

- "On-Demand" UAM
- City Taxi / Intra-City / Inter-City Transport as Main Use Cases
- Distributed Electric Propulsion — Quiet / Efficient / Clean / Safe
- Vertical Take-Off & Landing (VTOL) Aircraft — "Vertiports" / No Runways

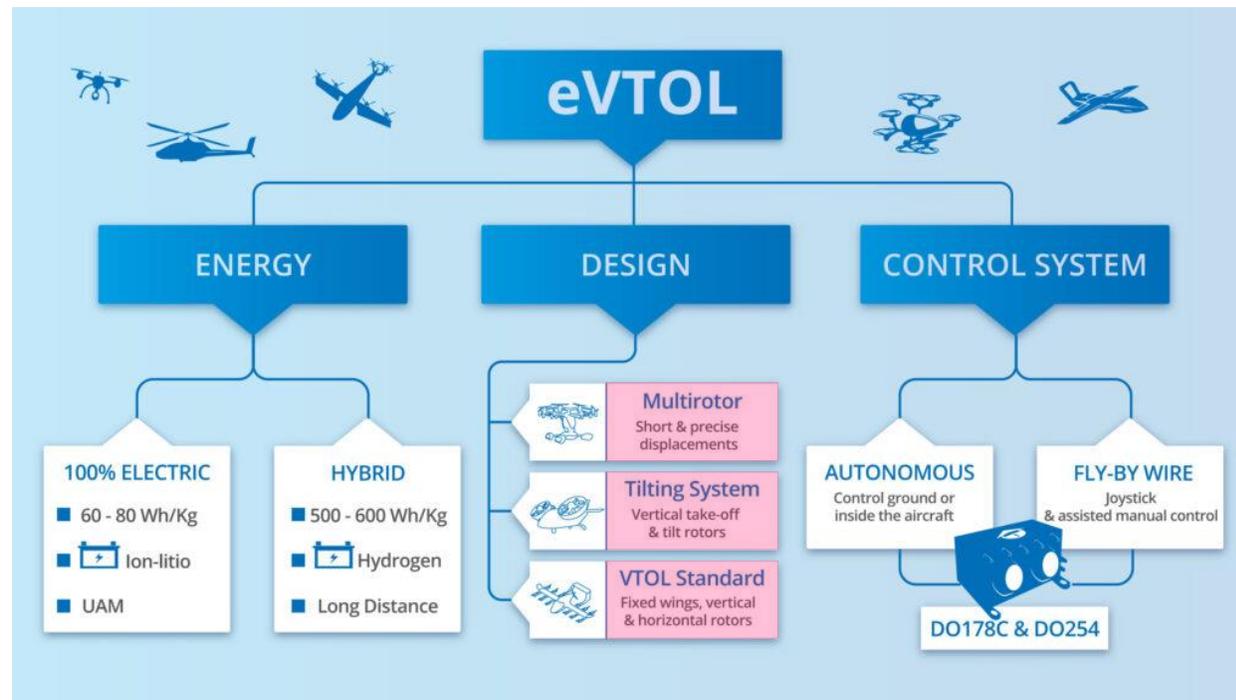
Source: Roland Berger 2020



- "Pop.Up Next" — Modular All-Electric Drone — 4x2 Rotors | People-Pod | EV Chassis (Airbus & Audi until 2019)

Types of eVTOL Aircraft

- **Multicopter** — Wingless / Distributes Thrust to Fly / Short Distances
- **Lift-Thrust** — Wings / Independent Lift & Thrust Units / Low Complexity
- **Vectored Thrust** — Wings / Propulsion Units Rotate to Provide Lift & Thrust



Source: www.embention.com

- **All-Electric Energy Supply** — Battery or Hybrid Battery / Fuel-Cell Combination

eVTOL Aircraft Concepts (1)

- **Volocopter (Germany) — VoloCity**
- **18 El. Rotors | Vert. Lift & Hover Flight | 900 kg Max. Take-Off Weight**
- **2 Passengers | 110km/h Cruise Speed | 36 km Range**

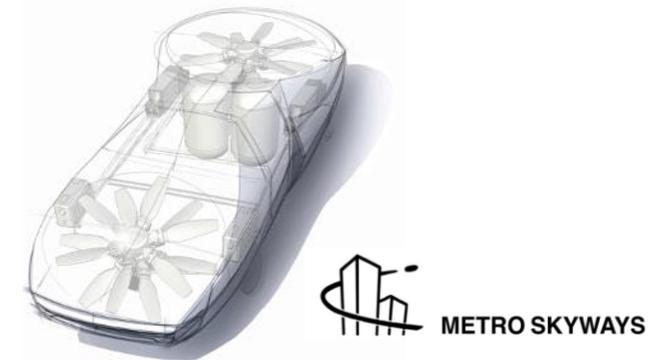


 VOLOCOPTER

- **EASA Certification Process On-Going (Target: 2024)**
- **Freight-Carrying *VoloDrone* Announced**

eVTOL Aircraft Concepts (2)

- Metro Skyways (former Urban Aeronautics Ltd., Israel) — CityHawk
- 2 Slow-Turning Ducted Fans | Vert. Lift & Horiz. Flight | 2000 kg Max. Take-Off Weight
- 4 Passengers | 280 km/h Cruise Speed | 280 km Range



- Initially Powered with Fossil Fuel | Transition to Fuel-Cell Power Supply
- Small Operating Space | Intended for Urban Areas / In Service 2028 - 2030

eVTOL Aircraft Concepts (3)

- **Joby Aviation (USA) — Joby S4 2.0**
- **6 Tilt-Propellers | Vert. Lift & Horiz. Flight | 2200kg Max. Take-Off Weight**
- **4 Passengers | 320km/h Cruise Speed | 240km Range**

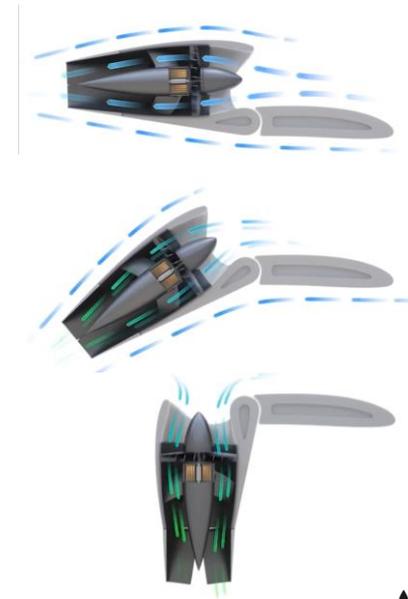


JOBY
AVIATION

- **Uber Elevate Acquired by Joby Aviation in 2020 / Commercial Operation Planned for 2024**
- **Battery Powered | Range Extension w/ Hybrid Fuel-Cell/Battery Architecture Announced**

eVTOL Aircraft Concepts (4)

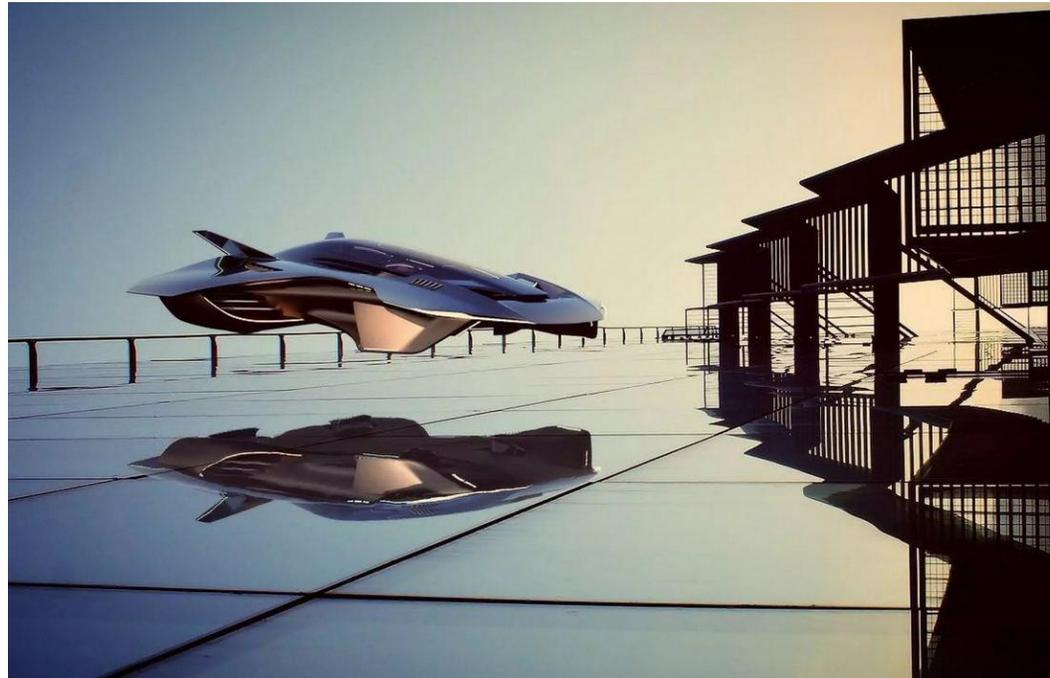
- **Lilium (Germany) — Lilium Jet**
- **36 Ducted El. Fans | Vectored Thrust — Vert. Lift & Horiz. Flight | 3100kg Max. Take-Off Weight**
- **6 Passengers | 280km/h Cruise Speed | 250km Range**



- **Partnership w/ Lufthansa Aviation / Commercial Operation Planned for 2025**
- **Extreme Requirements on Battery Technology / 320 kW Total Propulsion Power**

eVTOL Aircraft Concepts (5)

- **Bellwether Industries (UK) — Volar Antelope**
- **4 Ducted El. Fans | Hidden Propulsion System | 600 kg Max. Take-Off Weight**
- **1-2 Passengers | 220 km/h Cruise Speed | 330 km Range**



BELLWETHER
INDUSTRIES 

- **Half-Scale Demonstrator Tested**
- **Full-Scale Production Aircraft Planned with 4-5 Passengers**

eVTOL Aircraft Concepts (6)

- Skydrive Inc. (Japan) — SkyDrive SD-03
- 4x2 El. Rotors | Vert. Lift & Hover Flight | 400 kg Max. Take-Off Weight
- 1 Passenger | 50 km/h Cruise Speed | < 10 km Range



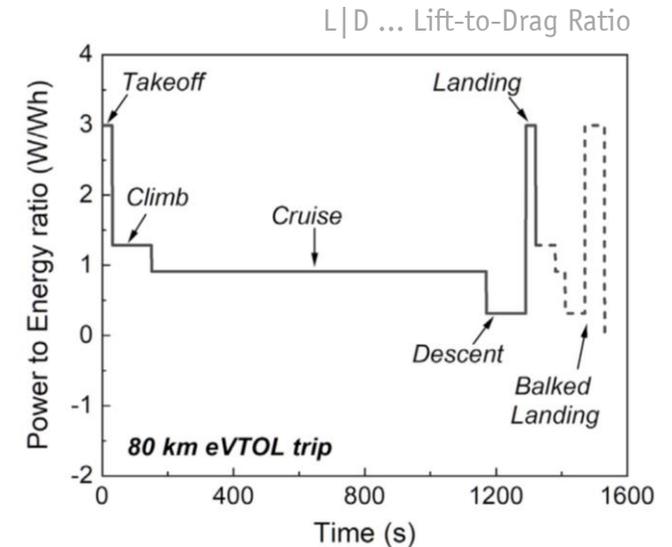
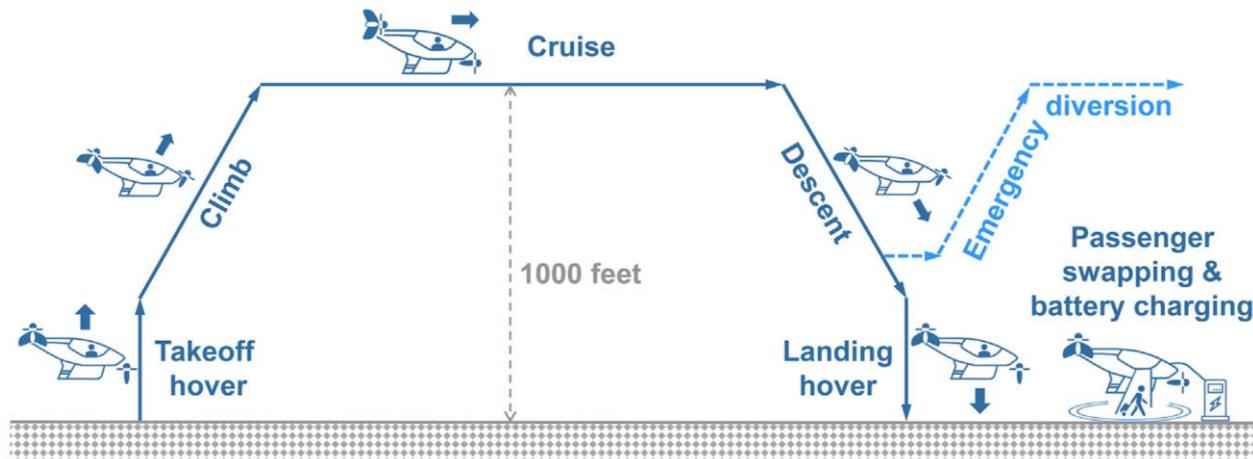
- Successful Manned Test Flight in 2020 / Type Certification Planned for 2025
- Mass Production of 30 kg Payload Cargo Drones Planned

eVTOL Mission Profile

— *Operational Requirements* —
Power / Energy Sources

Urban Air Mobility — Mission Profile

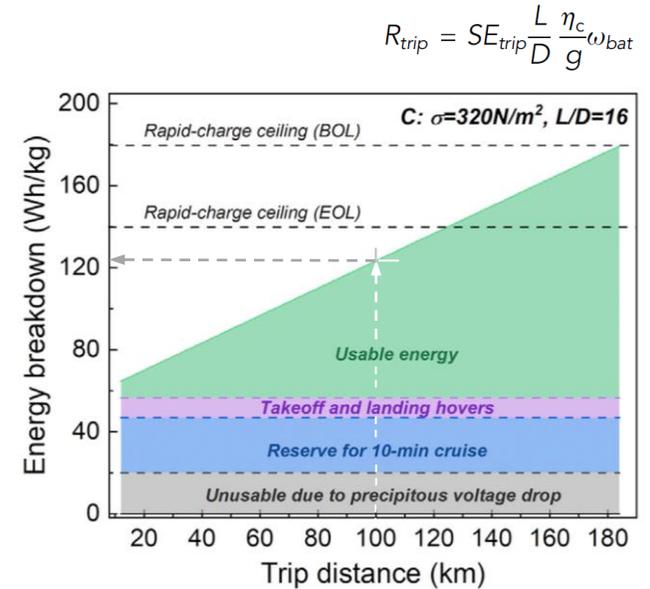
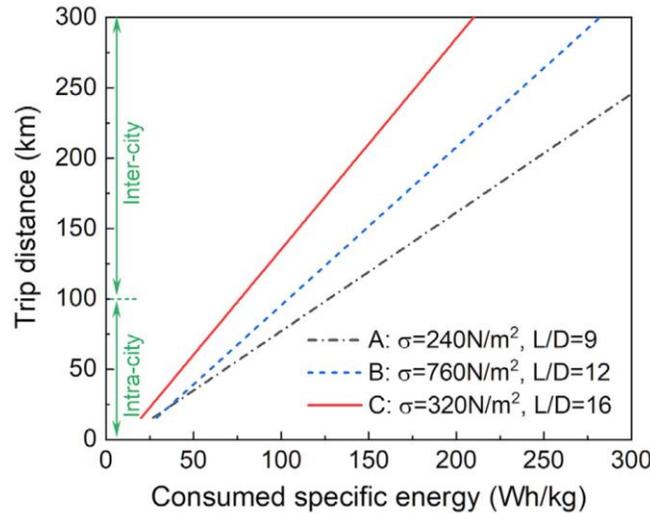
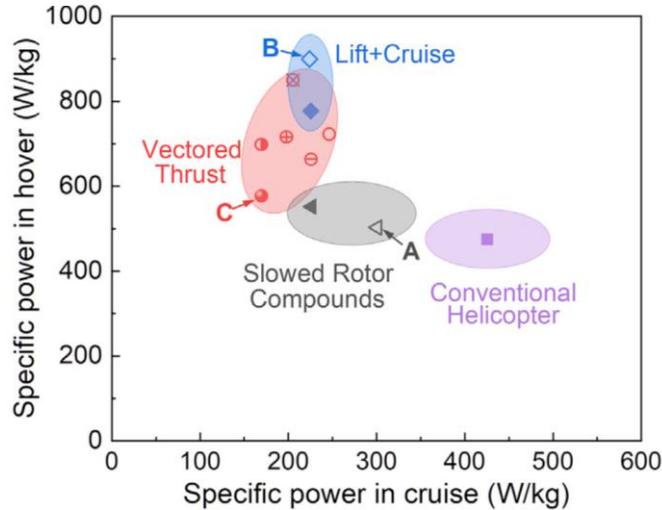
- **Multirotor eVTOL** — Large Disk Actuator → Low Disk Loading σ → High Eff. Hover / Low L|D → Low Cruise Eff.
- **Vecored Thrust eVTOL** — Wings → High L|D → Eff. Cruise / Low Hover Efficiency



- **Large Range** — High Spec. Energy Battery
- **High Payload** — High Spec. Power / High C-Rate Battery
- **High Vehicle Utiliz.** | Low Batt. \$\$\$ / Small Batt. — Fast High-Power Charging / High C-Rate / High # Cycles

Battery — Operational Requirements

- **Multicopter eVTOL** — Large Disk Actuator → Low Disk Loading σ → High Eff. Hover / Low L|D → Low Cruise Eff.
- **Vecored Thrust eVTOL** — Wings → High L|D → Eff. Cruise / Low Hover Efficiency



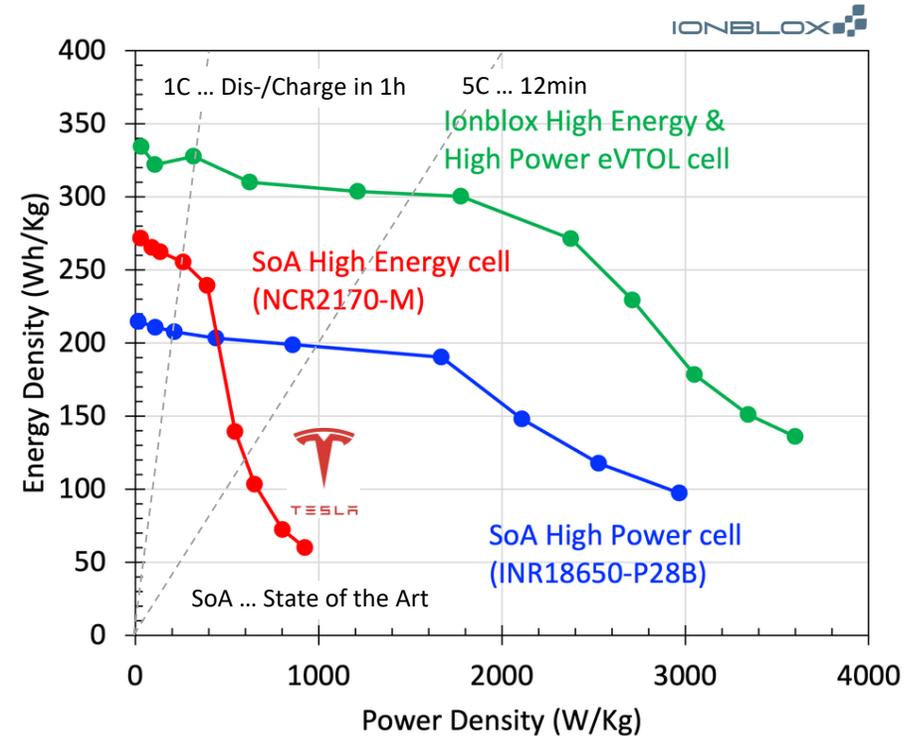
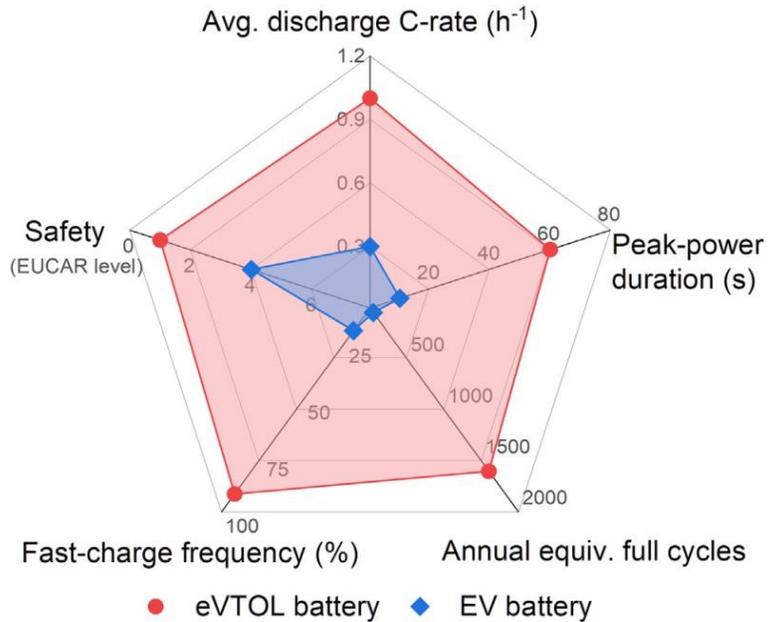
$$R_{\text{trip}} = SE_{\text{trip}} \frac{L}{D} \frac{\eta_c}{g} \omega_{\text{bat}}$$

- **Large Range** — High Spec. Energy Battery
- **High Payload** — High Spec. Power / High C-Rate Battery
- **High Vehicle Utiliz.** | Low Batt. \$\$\$ / Small Batt. — Fast High-Power Charging / High C-Rate / High # Cycles

$$t_{\text{char}} = \frac{SE_{\text{trip}}}{SP_{\text{char}}} = \frac{R_{\text{trip}}}{SP_{\text{char}}} \frac{g}{\eta_c \omega_{\text{bat}} L/D}$$

Battery Technology

- The “AND”-Challenge — High Specific Power & High Spec. Energy & High C-Rate & High Cycle Life
- Technology Interrelationships btw. Spec. Power / Spec. Energy / C-Rate (typ. 5C) / Cycle Life (typ. >1000)
- Battery Pack Wh/kg — typ. 80...90% of Cell



- Energy/Power Density Affects Payload & Range — Far Higher Requirements Compared to EV
- H₂ Fuel-Cells — typ. 500...1500Wh/kg | 400...600W/kg - Dependent on Payload & Range

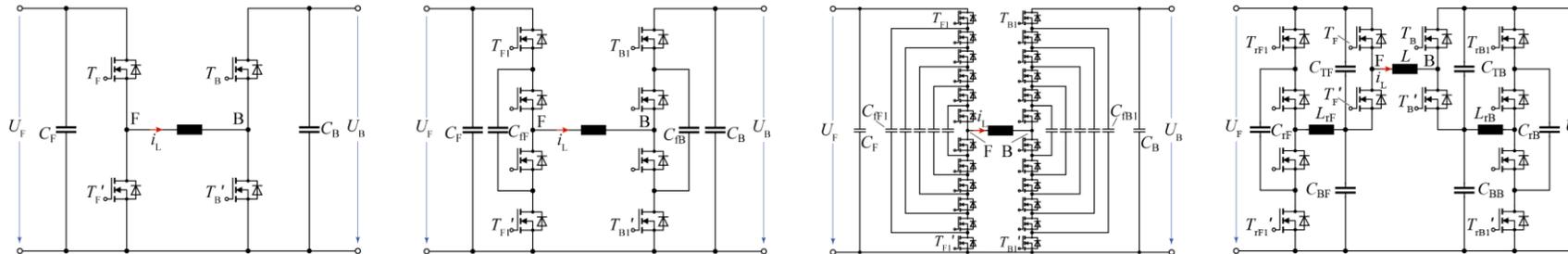
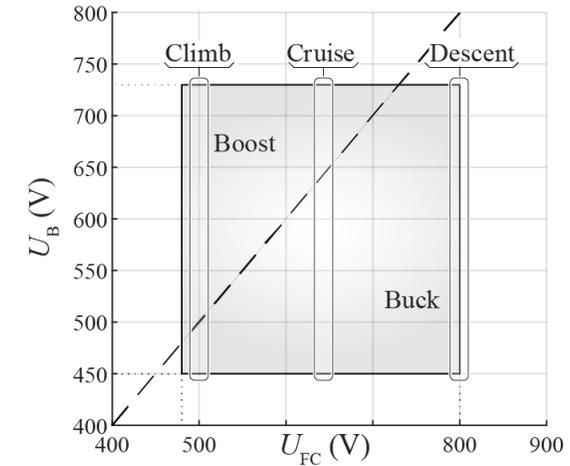
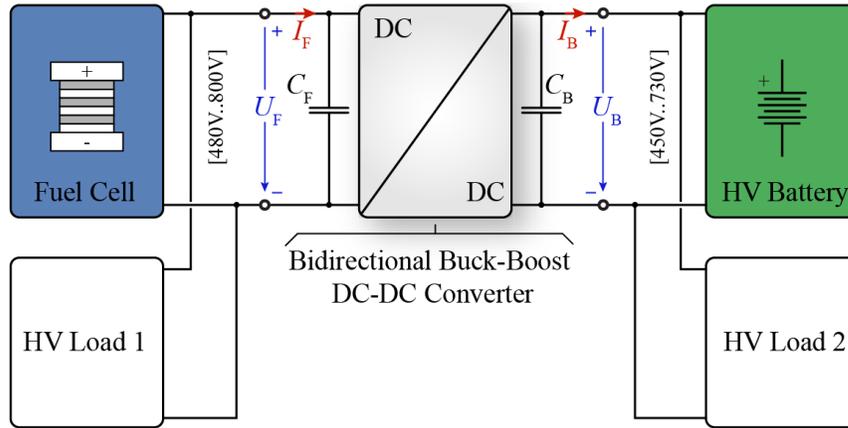


Ultra-Light Weight Power Electronics

*Electric Power System
Buck-Boost DC/DC Conversion*

Fuel-Cell/Battery Power Electronics Interface

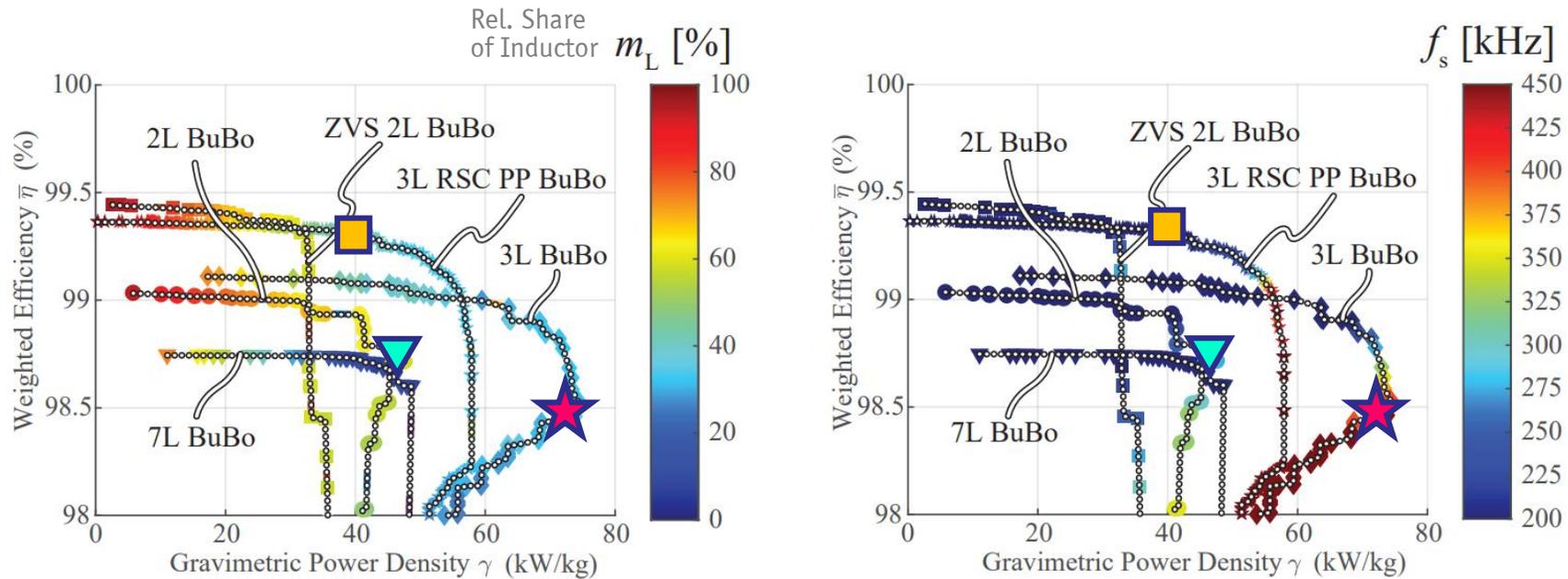
- Overlapping Input / Output Voltage Ranges — **Buck-Boost DC/DC Converter**
- Design Example — **Modular 150kW System** — **15kW Module** | $U_{FC} = 480...800V$ | $U_{Batt} = 450...730V$



- Multi-Objective Comparative Analysis — **2-Level (ZVS)** | **Multi-Level** | **Partial Power Processing Topology**

Fuel-Cell/Battery Power Electronics Interface

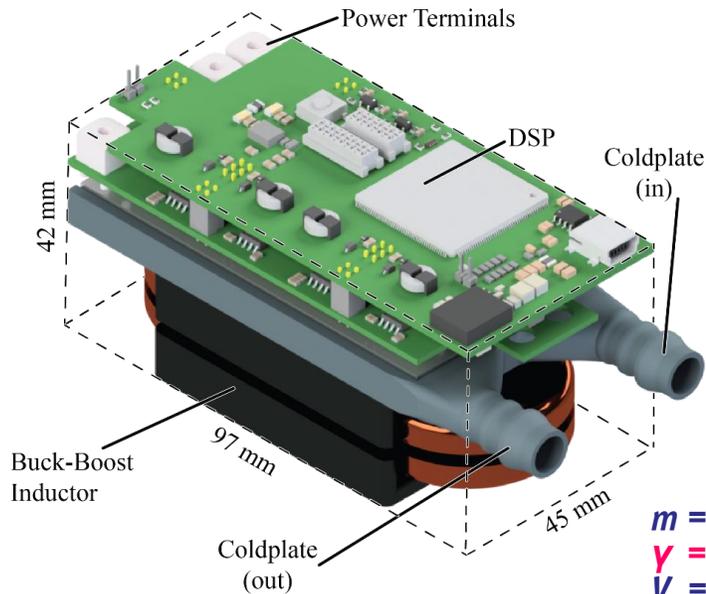
- Overlapping Input / Output Voltage Ranges — **Buck-Boost DC/DC Converter**
- Module — 15 kW | $U_{FC} = 480...800V$ | $U_{Batt} = 450...730V$



- **Multi-Objective Comp. Analysis** — 2-Level (SiC) $f_{sw} = 275$ kHz | 3-Level (GaN) $f_{sw} = 400$ kHz ($f_{sw,eff} = 800$ kHz)
- **Mission Efficiency** — 50% Climbing / 50% Cruising / No Fuel-Cell Power During Descent – DC/DC Conv. Off
- **System Considerations** — Battery & Fuel-Cell Weight vs. Converter Weight – Adv. of High Eff. Converter

Comparative Evaluation — 2L vs. 3L Converter

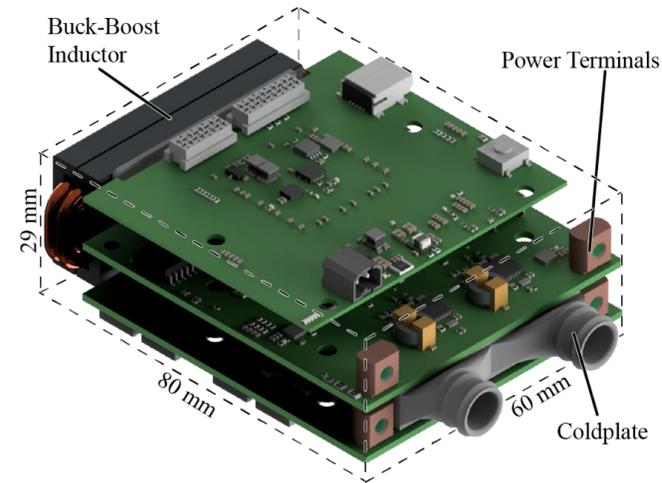
- Overlapping Input / Output Voltage Ranges — Buck-Boost DC/DC Converter
- Module — 15 kW | $U_{FC} = 480...800V$ | $U_{Batt} = 450...730V$



2-Level Converter

$m = 346\text{ g}$
 $\gamma = 43\text{ kW/kg}$
 $V = 183\text{ cm}^3$
 $\rho = 82\text{ kW/dm}^3$
 $\eta = 98.8\%$ (Mission)

2x



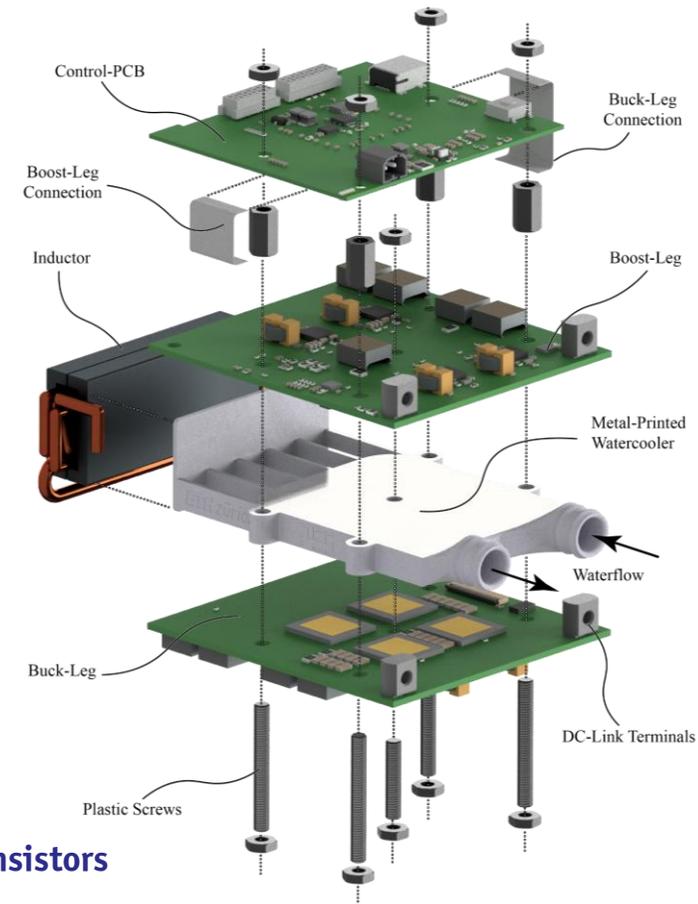
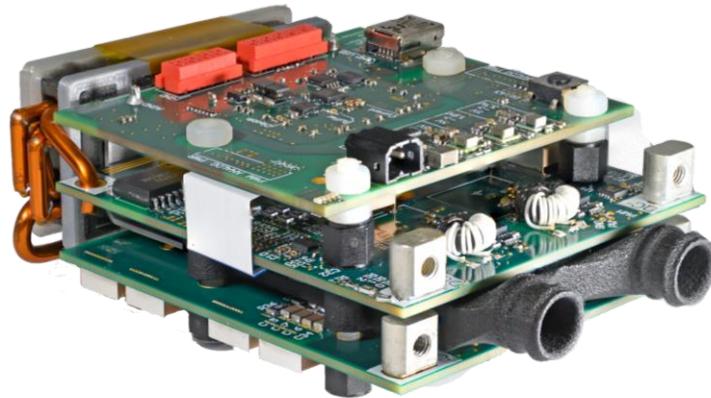
3-Level Converter

$m = 175\text{ g}$
 $\gamma = 86\text{ kW/kg}$
 $V = 139\text{ cm}^3$
 $\rho = 108\text{ kW/dm}^3$
 $\eta = 98.5\%$ (Mission)

- Exp. System Power Density Higher than Calculated due to 3D-Printed Cold Plate & Sandwich Structure
- Ultra-High Power Density — Design Target of 20 kW/kg Achievable w/ Low-Complexity 2-Level Approach

3-Level Converter — Hardware Demonstrator

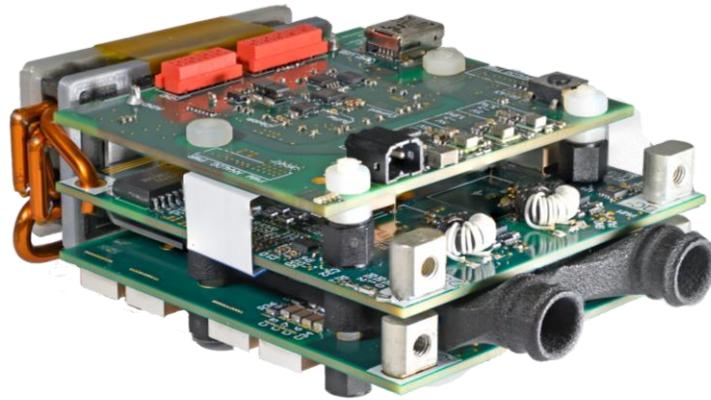
- Overlapping Input / Output Voltage Ranges — Buck-Boost DC/DC Converter
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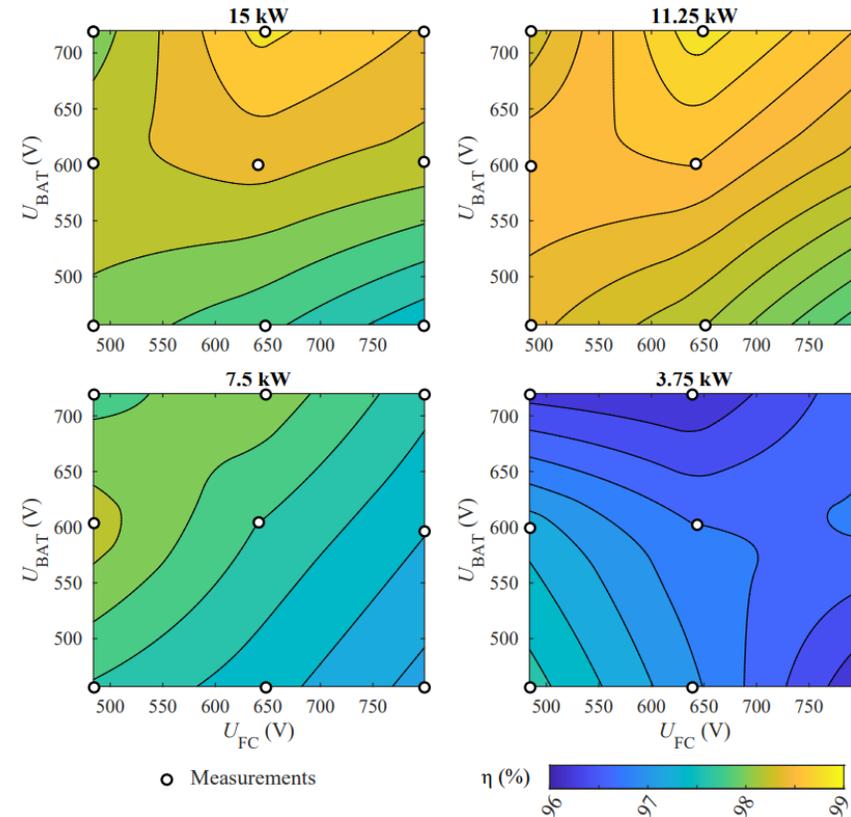
- 3D-Printed Alumina H₂O-Cooler / 600V GaN Power Transistors

3-Level Converter — Experimental Results

- Overlapping Input / Output Voltage Ranges — Buck-Boost DC/DC Converter
- Module — 15 kW | $U_{FC} = 480 \dots 800 \text{ V}$ | $U_{Batt} = 450 \dots 730 \text{ V}$



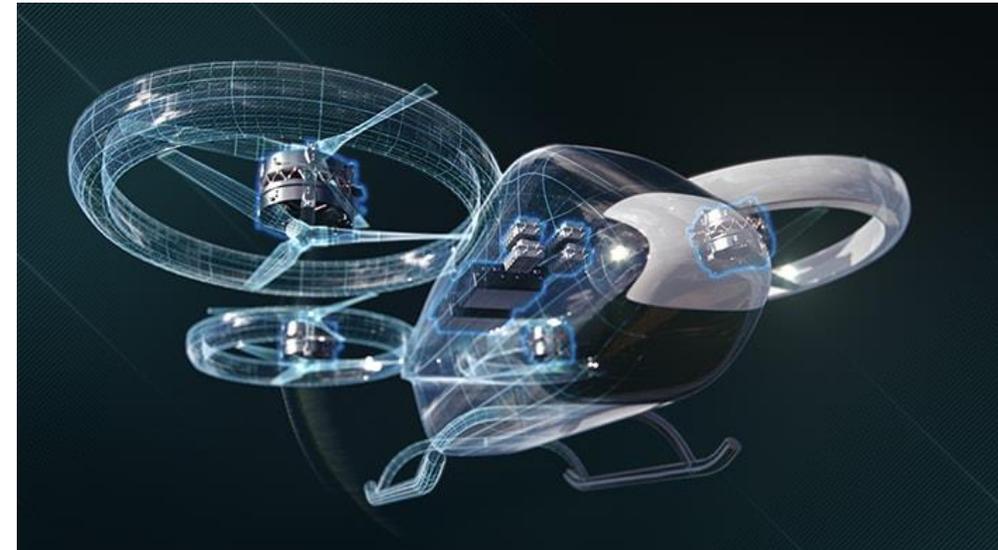
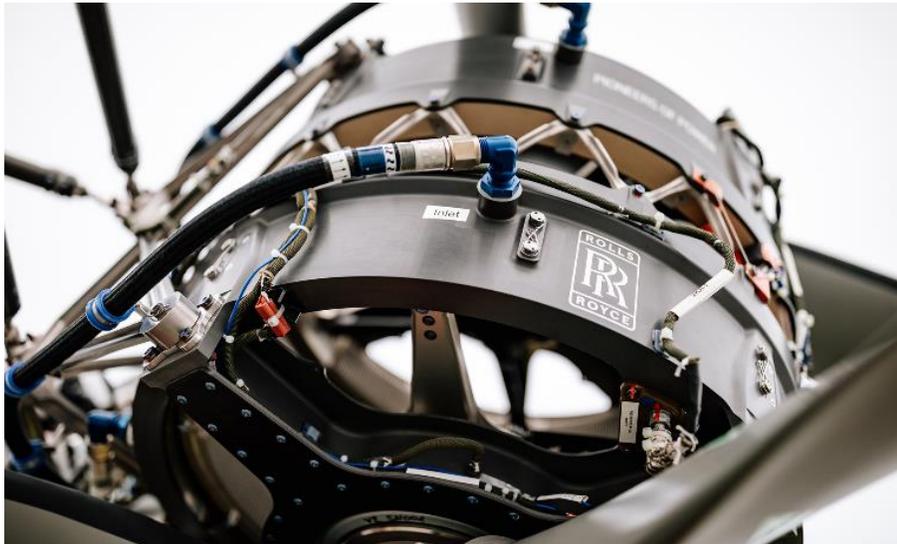
★ $\gamma = 86 \text{ kW/kg}$ @ $\eta_{max} = 98.9\%$



eVTOL Electric Motor Technology

- Best-in-Class — 30Nm/kg | 15kW/kg
- Motor Scaling Acc. to Torque M (!) | $P = M \cdot \omega$ – High Speed/Power Density & Gearbox OR Direct-Drive
- Adv. / Low-Weight Radial- or Axial-Flux Motors incl. Concepts w/ Integ. Magnetic Gear — 20kW/kg Target

Source: AIRBUS



- CityAirbus Demonstrator (2020) — 8x 100kW Direct-Drive Rolls-Royce (SIEMENS) Adv. Motor Technology
- 4x2 Ducted Co-Axial Propellers – Low Noise | 400kg Trust / Duct | 250kg Payload | 120km/h for 15 min

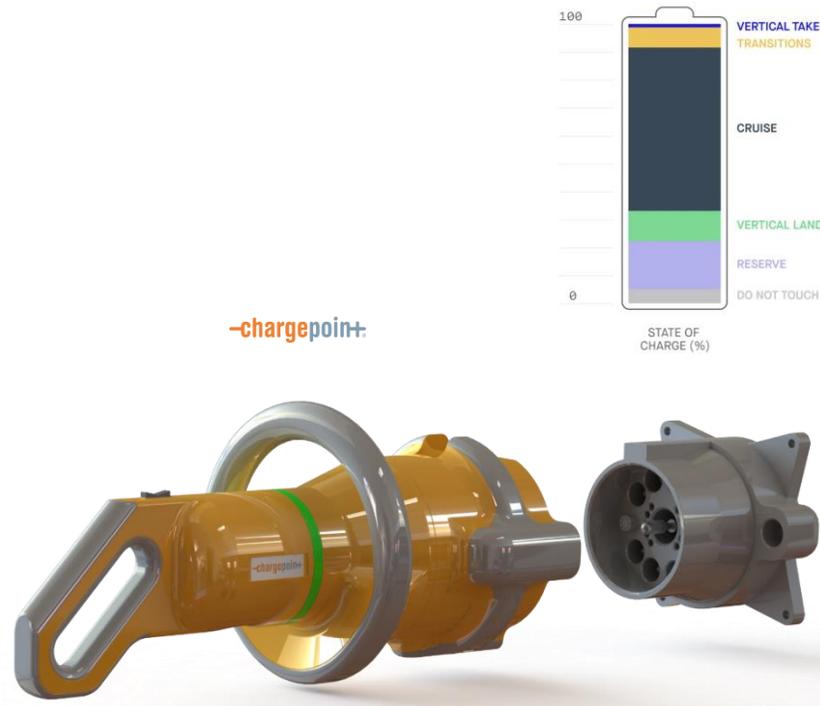
High-Power Battery Charging

————— *50Hz Transformer & Active Rectifier* —————
Solid-State Transformers

Up to 3.5 MW Ultra-Charging System

- 20% → 100% **Charging of typ. 200...400kWh Battery Pack in 15...20 min** — High eVTOL Utilization
- **MegaWatt Charging System (MCS)** — New Charging Std. for Trucks | Buses | eVTOL Aircraft etc.

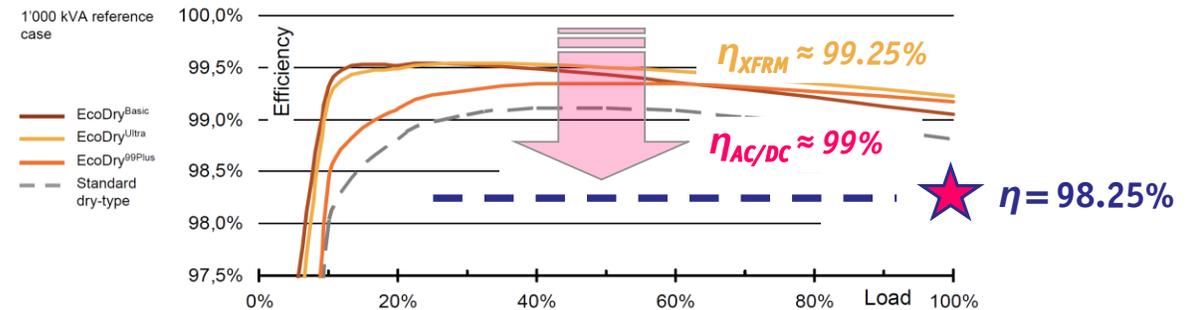
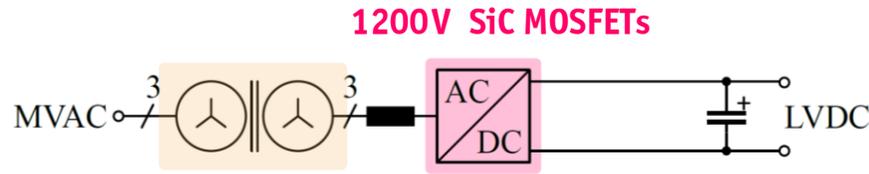
Source: **NEW ATLAS**



- **ChargePoint High-Power Charge Connector (max. 1.5 kV/3000A) incl. Liquid Cooling & Data Transfer**

Dry-Type LFT Technology & SiC PFC Rectifier

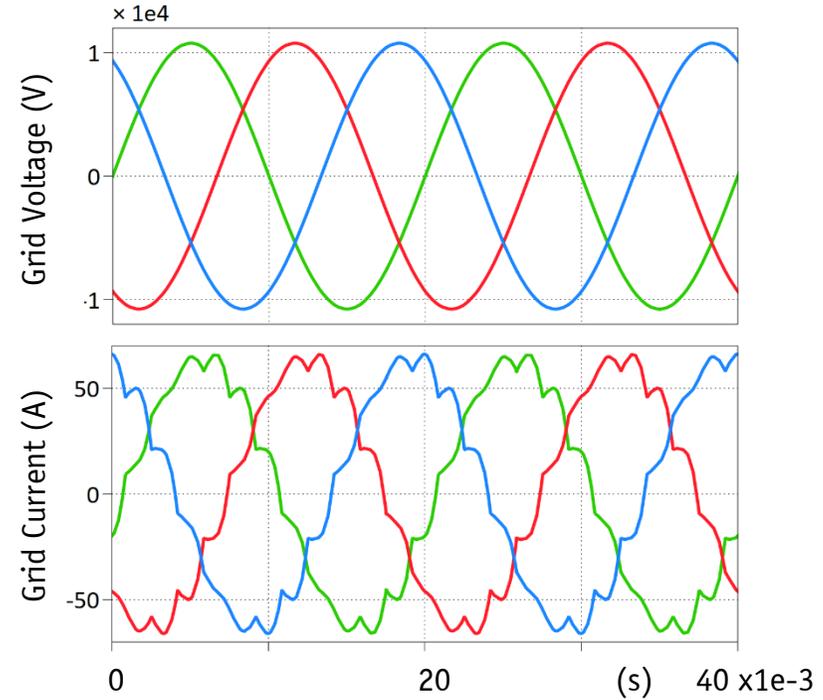
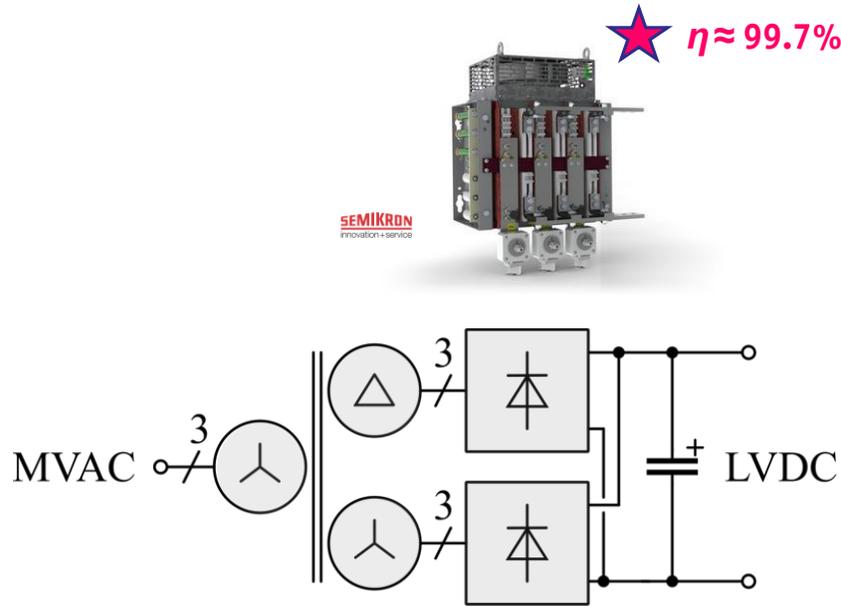
- 400 kVA EcoDry™ High-Efficiency Transformer
- Vacuum Cast Coils — No Fire Hazard
- Amorphous Metal Core — Low No-Load Losses
- High Overvoltage / Overload Capability



- Utilizing Proven LV SiC MOSFETs in AC/DC Stage → $\eta_{AC/DC} = 99+ \%$ Efficiency
- Full Functionality — Bidir. Power Flow | Scalability to Higher MVAC-Levels
- No DC Fault Current Limit (!)

3- Φ 12-Pulse / Multi-Pulse Rectifier

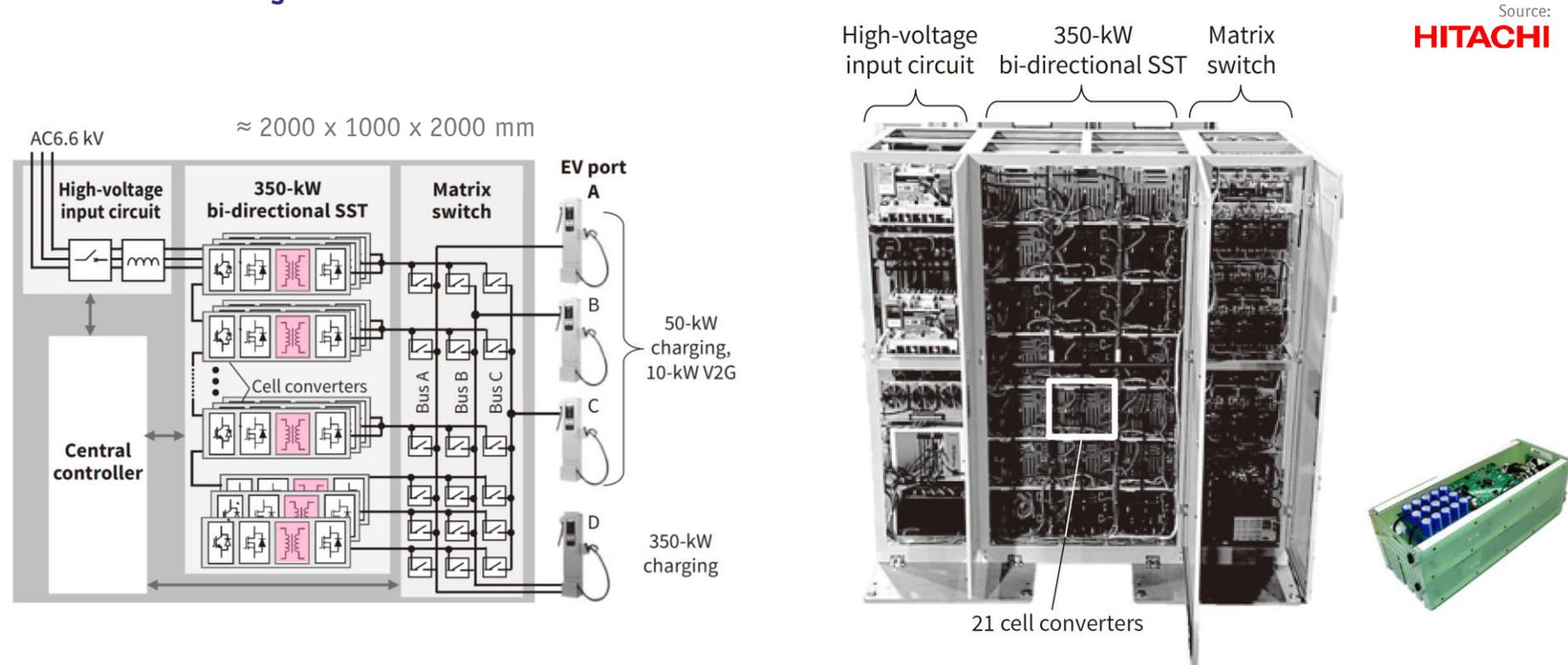
- **No Explicit PFC Stage (!)** — Passive Realization of PFC with Phase-Shifting Transformer / **No Inductors**
- **Low Complexity | High Robustness | Long Lifetime**
- **18-Pulse, 24-Pulse for High Power Levels**
- **4 kW/dm³ Rectifier Stage / Air Cooling**



- **Unidirectional**
- **No Active Output Voltage Control**
- **Remaining Current Distortion / Reactive Power Consumption**

3- Φ 6.6kV / 350kW SST-Based EV Charger (1)

- 3x7 = 21 Cells | 5 kHz 1.7 kV Si-IGBT AC/DC Stage | 50 kHz 1.7 kV SiC 1050V/400V DC/DC Converter
- Matrix Switch Output for 21x 17 kW \rightarrow 1x 350 kW Charging Port Config. & Cascaded Cell Balancing
- Forced Air Cooling



- Power Density \rightarrow 0.09 kW/dm³ (System) | \approx 0.18 kW/dm³ (SST/Cells incl. Isol.)
- -40% Footprint / -70% Weight vs. LFT-Based Solution / 83% Lower Transf. Volume

3- Φ 6.6kV / 350kW SST-Based EV Charger (2)

- 3x7 = 21 Cells | 5 kHz 1.7 kV Si-IGBT AC/DC Stage | 50 kHz 1.7 kV SiC 1050V/400V DC/DC Converter
- Matrix Switch Output for 21x 17 kW \rightarrow 1x 350 kW Charging Port Config. & Cascaded Cell Balancing
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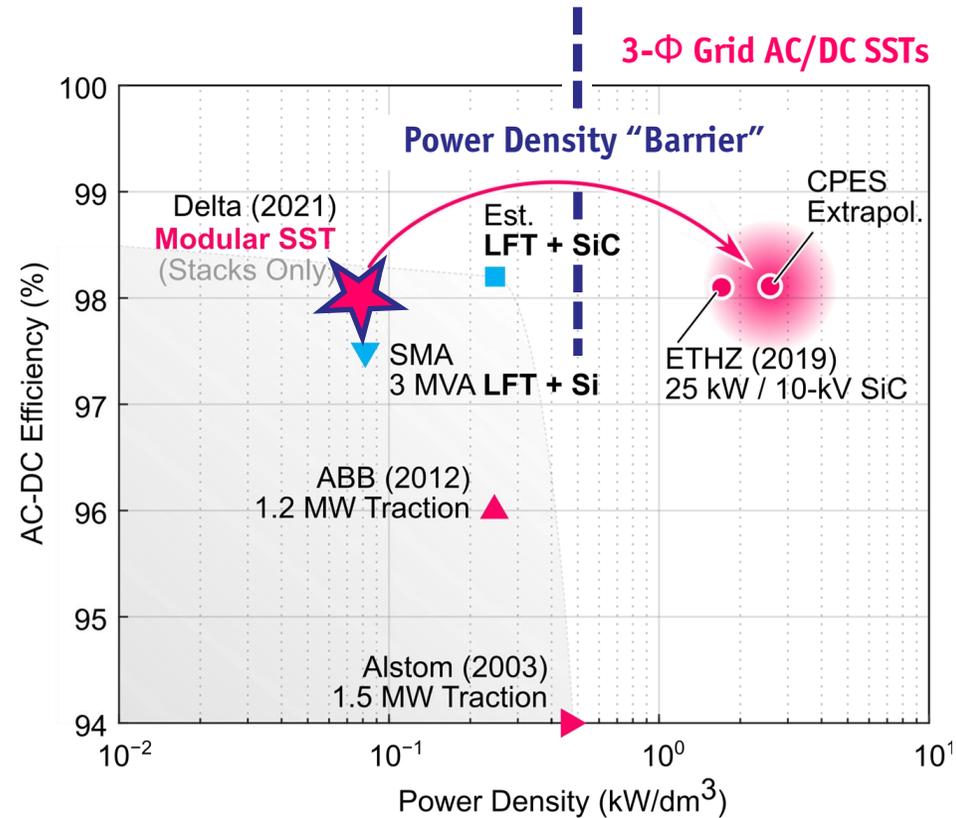


Source:
HITACHI

- Power Density \rightarrow 0.09 kW/dm³ (System) | \approx 0.18 kW/dm³ (SST/Cells incl. Isol.)
- -40% Footprint / -70% Weight vs. LFT-Based Solution / 83% Lower Transf. Volume

Next.-Gen. SSTs — Research Targets

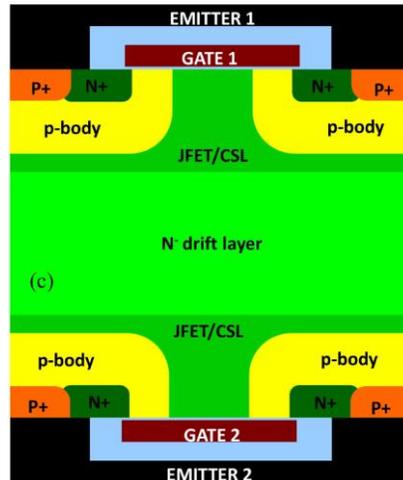
- 10+ kV SiC and/or New Topologies Might Facilitate to Overcome the Power Density Barrier
- AC/DC Efficiencies >98% Remain Difficult to Attain



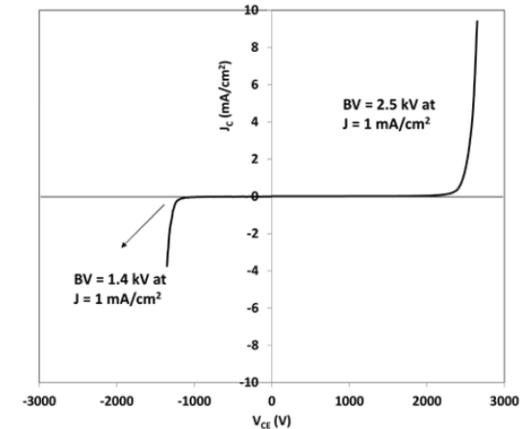
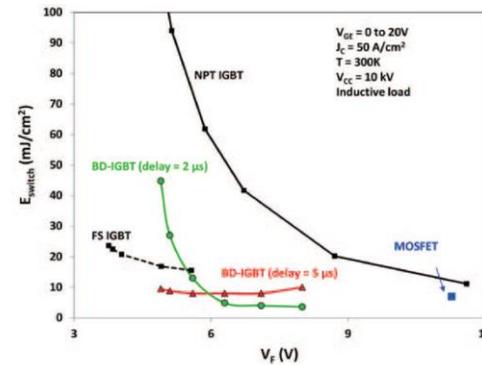
- Full-Scale Demonstrators incl. Mains-Side Overvoltage / DC-Side Overcurrent Protection (!)

Next.-Gen. SSTs — Monolithic Bi-Direct. 15 kV SiC IGBT

- Planar-Gate Bi-Direct. IGBT Fabricated w/ Double-Sided Lithography Process
- Conduction & Sw. Loss Influence of Back-Side Gate Voltage Bias
- Challenging Packaging & Cooling



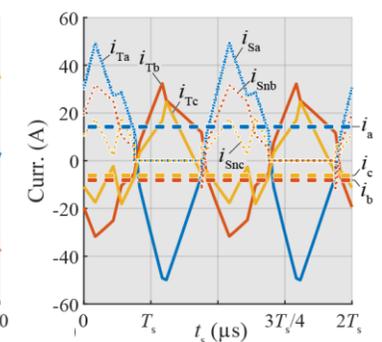
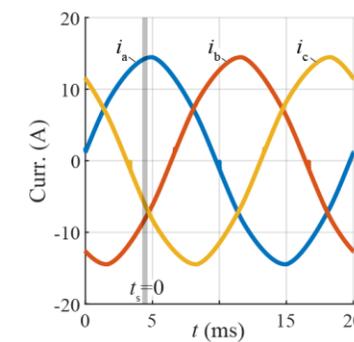
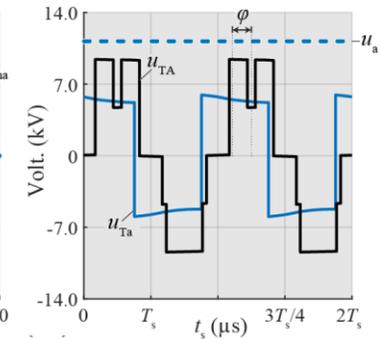
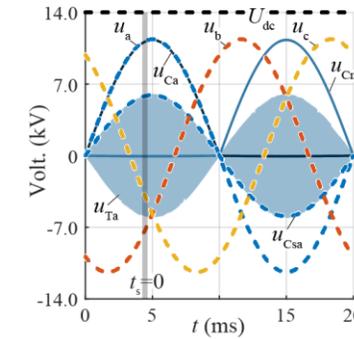
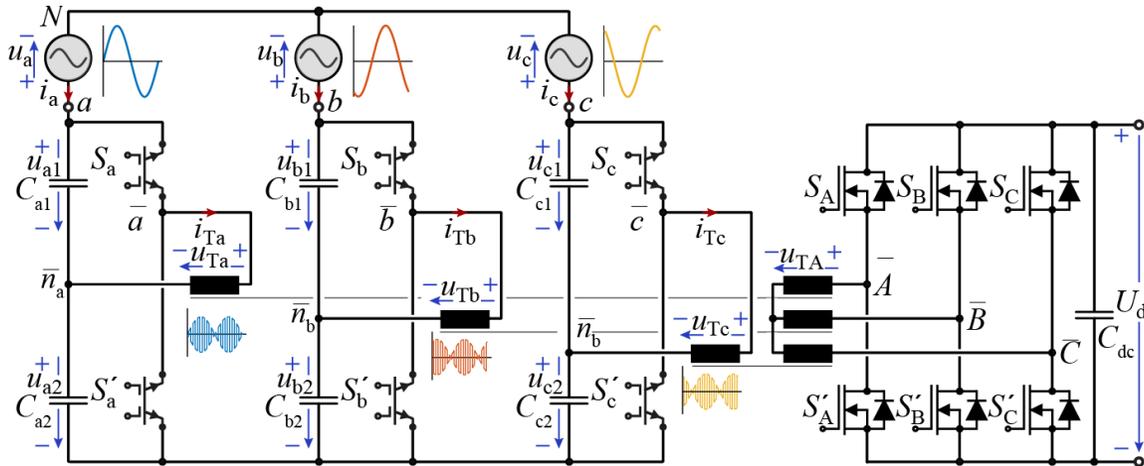
Chow et al., 2016



- Simul. Performance of a 15 kV BD-IGBT | Max. 7.2 kV Measured — Epi Layer Defects etc.

Next.-Gen. SSTs — *Novel Swiss-SST Concept*

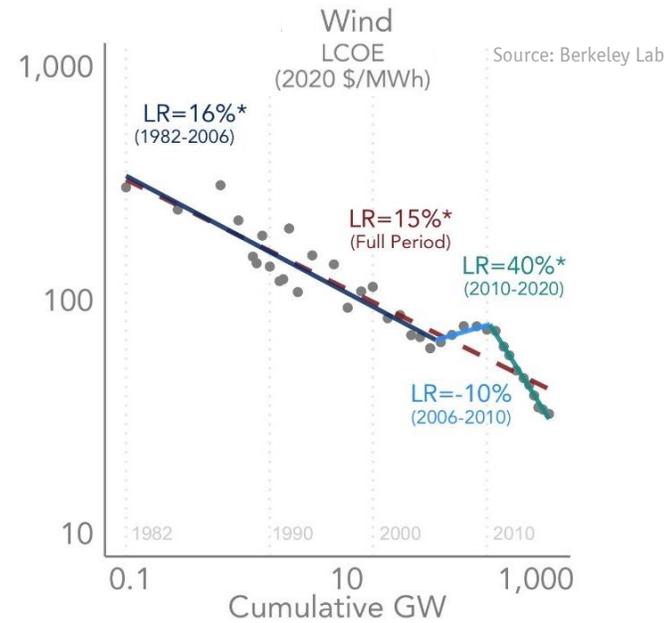
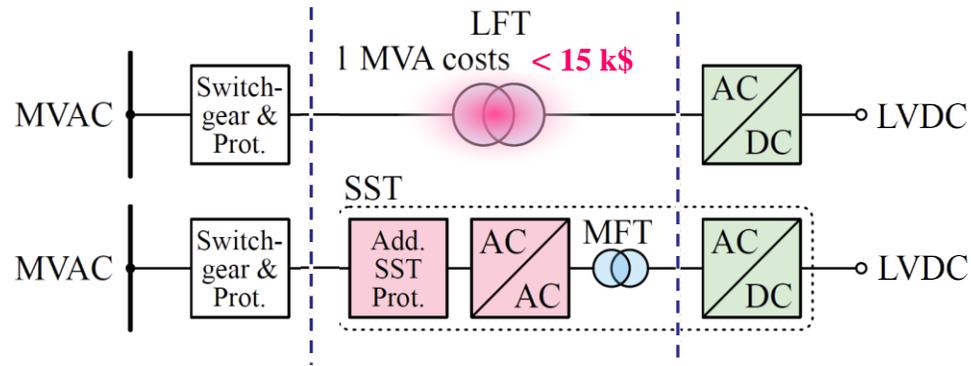
- **“Isolated Front-End”-Topology** — Minimum Complexity of MV Power Electronics & Control Circuit
- **15 kV SiC MBD-IGBT** — Direct Connection to 13.2 kV Distribution Grid
- **Unity Power Factor / Bidirectional**



- **Dual Active Bridge-Type Control**
- **AC-Side Phase Modularity** — Full Rated Power Operation @ 1-Φ Input (!)

Outlook — SST Technology Learning Curve

- **Learning Rate** — Cost Reduction for Each Doubling of the Cumulative Production
- **Prediction** of Future Costs of a Technology (e.g. PV or Wind “Grid Parity”)



- **Typ. Empirical Learning Rates of 15...25%** → Dramatic Cost Reduction Over Longer Timespan
- **15 k\$ Budget for 1 MVA SST MV Power Electronics & MFT ≈ 10 k\$/1 MVA Power Converter** → **10 \$/kW (!)**

! Remark — *Battery Swapping*

- Exchange of Drained Battery in **Only 5min (!)**
- Re-Charging of Batteries in Controlled Environment @ Low Energy \$\$\$ Time Periods

Source: <https://wassafss.medium.com>



- **Disadvantage of Standardized Single-Pack Battery Design & Battery Accessibility**

Economic Perspectives & Sustainability

*Market Growth Perspectives
Carbon Footprint vs. ICE & EV*

Urban Air Mobility (UAM) Market Forecast

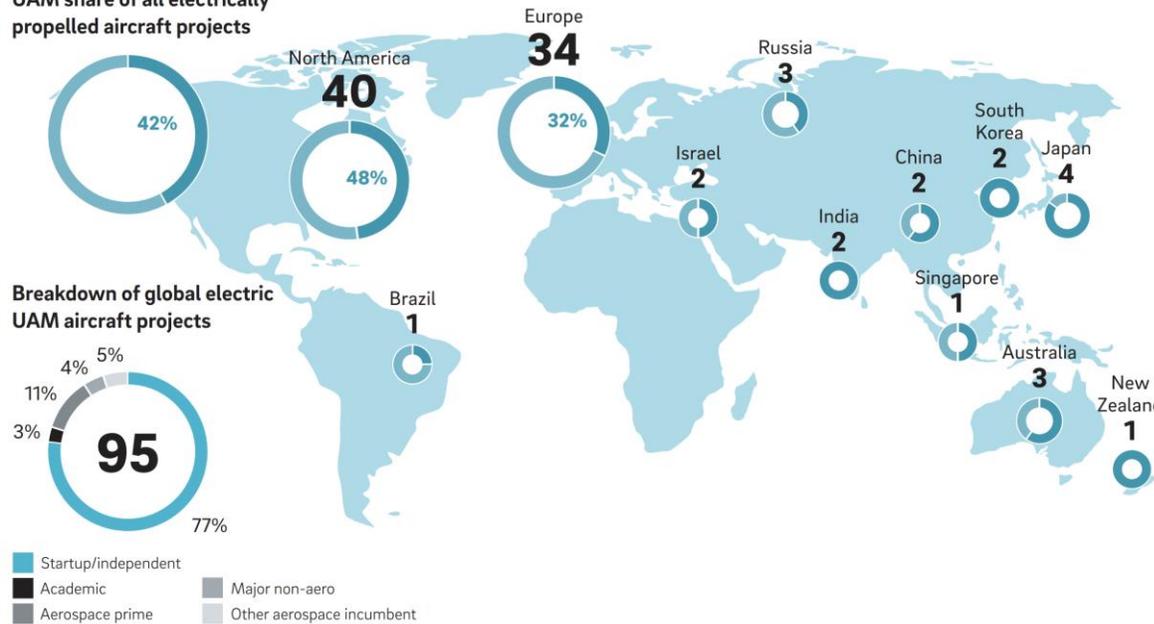
- **160'000 Passenger Drones** Expected by **2050** — Add. Market for Services / Repairs etc.
- **USD ≈900 Million Investments** in 1st Half of 2020 — 20x Level of 2016 (Full Year)

The Global Electric Propulsion Radar

Distribution of the 95 known electric UAM aircraft projects

Source:  2020

UAM share of all electrically propelled aircraft projects



- Industry Expected to be Ready for Take-Off in 2025
- By 2050 Estimated Revenues of USD ≈90 Billion/Year (≈1 Billion in 2030)

Certification & Future Airspace Interaction Concept

- US Federal Aviation Admin. (FAA) / EU Aviation Safety Agency (EASA) — Regulations & Certifications
- Buildings / Towers & Noise-Sensitive Areas — Def. of Low-Altitude UAM Corridors & Holding Areas

Source: Y. Lee et al. | applied sciences



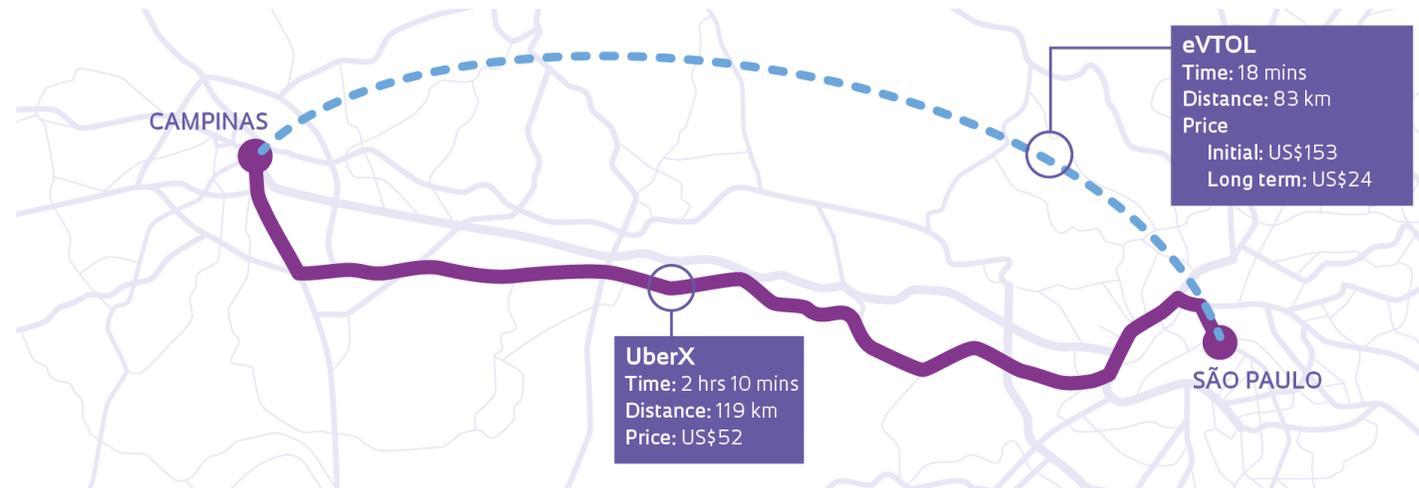
 **VOLOCOPTER** — Targets EASA Certification for “VoloCity” in 2024

UAM Comparative Evaluation (1)

- Study of **UBER Elevate (2015)** on Cost / Time of Commuting btw Cities — **eVTOL Aircraft vs. Cars**
- «On-Demand» Urban Air Transportation — UberCopter as 1st Step

Car vs. eVTOL ▲

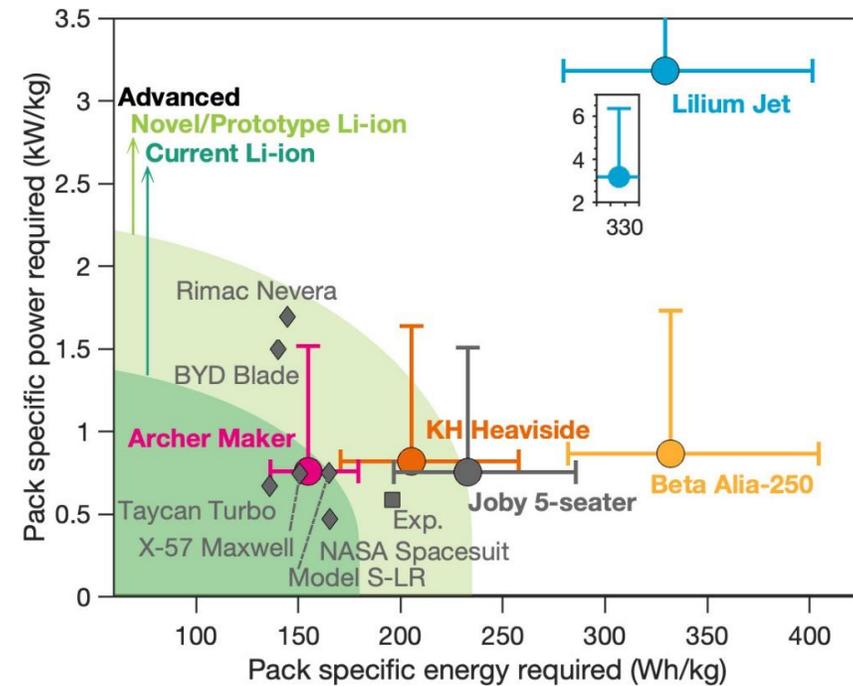
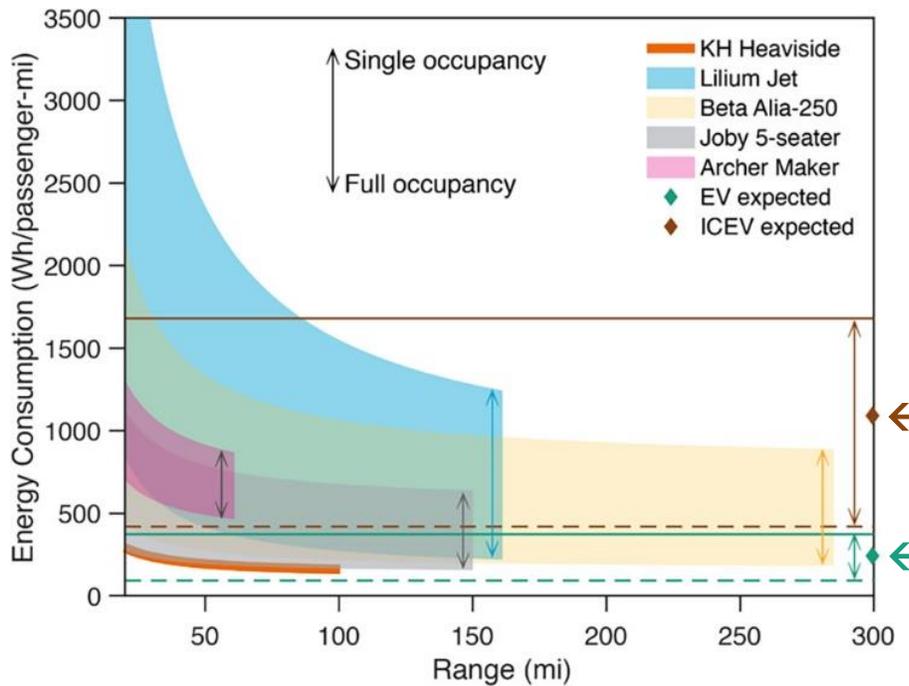
Uber study compared the cost and time of commuting between the cities of São Paulo and Campinas by car and eVTOL



- **Lilium** — 6 Passengers | 250km Range | 280km/h Cruise Speed | 20-25 Flights per Day

UAM Comparative Evaluation (2)

- eVTOL Aircraft Provide 2x ... 6x Faster Means of Point-to-Point Mobility
- Up to 300 mi of Range with up to 7 Passengers Using Latest Battery Technology



- EV and ICEV — 220 Wh/Passenger-mi and 1,000 Wh/Passenger-mi
- eVTOL Aircraft — 130 Wh/Passenger-mi ... ≈ 1,200 Wh/Passenger-mi Dep. on Design & Occupancy

Thank you!



Source: Paul Bunch
<https://leslikely.artstation.com/>